



Research article

Nonlinear robust control of trajectory-following for autonomous ground electric vehicles with active front steering system

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Abstract: This paper presents a nonlinear robust H -infinity control strategy for improving trajectory following performance of autonomous ground electric vehicles (AGEV) with active front steering system. Since vehicle trajectory dynamics inherently influenced by various driving maneuvers and road conditions, the main objective is to deal with the trajectory following control challenges of parametric uncertainties, system nonlinearities, and external disturbance. The AGEV system dynamics and its uncertain vehicle trajectory following system are first modeled and constructed, in which parameter uncertainties related to the physical limits of tire are considered and handled, then the control-oriented vehicle trajectory following augmented system with dynamic error is developed. The resulting nonlinear robust H -infinity state-feedback controller (NHC) of vehicle trajectory-following system is finally designed by H -infinity performance index and nonlinear compensation under AGEV system requirements, and solved utilizing a set of linear matrix inequalities derived from quadratic H -infinity performance and Lyapunov stability. Simulations for double lane change and serpentine scenes are carried out to verify the effectiveness of the proposed controller with a high-fidelity, CarSim[®], full-vehicle model. It is found from the results that the proposed NHC provides improved vehicle trajectory following performance compared with the linear quadratic regulator (LQR) controller and robust H -infinity state-feedback controller (RHC).

Keywords: autonomous vehicles; electric vehicles; trajectory-following; robust control; nonlinear control

Mathematics Subject Classification: 93B52, 93C95, 93D05

1. Introduction

The emergence of AGEV attracted academia research interest and attention due to it has great advantages in reducing road congestion, environmental pollution and improving traffic safety in recent years [1–4]. As the new technology of chassis active control for AGEV steer system, active front steering (AFS) is used to improve lateral stability and active safety by applying variable steering gear ratio in different speed scenarios. AGEV equipped with AFS systems can bring strong guarantees of driver safety and handling flexibility through quick input response and efficient steering execution, it has been widely applied and developed in trajectory-following control system for autonomous vehicles [5–7]. In view of the advantages of fast response and accurate execution, AFS system has excellent abilities to enhance the active safety of vehicle and the following performance of target trajectory for AGEV [8–13].

The trajectory-following for AGEV with AFS system has been explored in many literatures [14,15]. The feedforward and feedback controller of the multi-rate Kalman filter with compensation ability is designed by considering the difference between the control period of the AGEV motor and the sampling time of the ordinary camera, and its AFS is used to realize the trajectory-following successfully [16]. Aiming at the mutual interference and control distribution problems of AFS system and electronic stability control system for AGEV, a novel integrated stability model predictive control (MPC) controller is proposed in [17], and the control of steering and braking based on tire force is reassigned to follow the target trajectory. In order to improve the handling stability and the steering comfort for AGEV, the work in [18] proposed a variable steering ratio AFS controller by constructing the mapping of steering wheel angle and longitudinal vehicle speed. Trajectory-following control in [19] is based on successive online linearization of the vehicle model of autonomous vehicles, and its controller computed the front steering angle of AFS to follow the trajectory on slippery roads. In addition, due to the advanced steering technology of the AFS system, it also plays a non-negligible role in other related application fields of trajectory-following control [20–22].

Although the aforementioned efforts of trajectory following were successful, there are still some challenges such as uncertain model parameters, system nonlinearity and external disturbances [23,24]. To address above challenges, nonlinear model predictive control (NMPC) is used to transform the nonlinear system into a parameter-dependent adjustment problem, its sufficient conditions for the feasibility and convergence are given in [25]. In order to reduce the influence of external disturbances on trajectory-following, a combination of super twisting second-order sliding mode (SOSM) and nonlinear disturbance observer technique is used to estimate the environmental disturbances and modeling errors, and system's robustness and security is gained through hardware-in-the-loop experiments [26]. A novel adaptive sliding mode control (SMC) frame is presented to solve the safety problems caused by actuator failures and external attacks of Markov jump cyber-physical systems, a new linear-type switching surface is established by state estimator, and the stability criterion of system with uncertainty is also deduced, and then the effectiveness of proposed control strategy is final tested in tunnel diode circuit experiments [27]. To accurately track the motion trajectory of the uncertain industrial robotic arm system, a SMC framework with switching neural networks is proposed in [28], and the adaptive algorithm technology and radial basis function are used to provide real-time feedback for the multi-modal system with control gain so that the robustness of the trajectory following control system can be enhanced. A novel composite adaptive fuzzy controller (FC) is proposed to solve the nonlinear problem of the system following model in [29], its strict stability analysis is carried out by

Lyapunov method, and the performance of the proposed controller is also verified. Some extensions of FC can be found here [30,31]. The active disturbance rejection control (ADRC) is also used to estimate and compensate the unmodeled dynamics and unknown external disturbances of the system in real time to improve the robustness of vehicle trajectory-following [32]. An optimal preview controller is designed to improve the dynamic trajectory following performance and high-speed motion stability for the articulated vehicle through the compound lateral position deviation, and the vehicle following performance and motion stability are verified in experiments [33]. An MPC strategy for optimal speed is presented to achieve fast and accurate trajectory tracking in [34], and the appropriate front wheel steering angle is calculated to obtain a satisfactory trajectory following effect, experimental results verify the effectiveness of the proposed strategy. An optimal control strategy with nonlinear time delay differential equation is used to suppress the chatter of the milling system in [35], and the optimal control rate is obtained with the extreme value change algorithm to optimize the chatter performance index of the system. The work [36] presented an adaptive multi-model switching control method to solve the parameter jump problem of complex nonlinear systems, and some key lemmas and extended recursive least squares algorithms are used in the design of adaptive control law. To deal with unstructured disturbances of nonlinear systems, a new nonlinear optimal control technique is proposed by iterative techniques in [37], and the conditions for asymptotic stability of the system and the corresponding H-infinity control law are given. The study [38] designed a new global adaptive control based on output feedback to deal with nonlinear and unknown parameters of the system, the proposed system feedback controller can ensure the convergence of the system state, boundedness of other signals by scaling technology and backstepping method. Meanwhile, robust control has been utilized in the field of trajectory-following control for its advantages in dealing with system nonlinearity [39], linear variation of parameters [40] and external disturbance of the system [41].

Therefore, this paper proposes nonlinear robust control strategy of AGEV with AFS to deal with the trajectory following control challenges of parametric uncertainties, system nonlinearities, and external disturbance. The AGEV system dynamics and its uncertain vehicle trajectory following system is first established. Then, under the requirements of H-infinity performance index, system nonlinear compensation and trajectory-following target, the nonlinear robust state feedback controller for AGEV of trajectory-following is designed. Finally, the effective of the proposed controller is verified with double lane change and serpentine scenes in Simulink/Carsim. The rest of paper structure is as follows: problem description and trajectory-following modeling are presented in Section 2. Section 3 provides the design process of nonlinear robust controller. Simulation results are analyzed and discussed in Section 4, and the conclusions are offered in Section 5.

2. Problem description and trajectory following model

2.1. Vehicle dynamics system model

The paper mainly studies the trajectory following of AGEV problem, we assume that the suspension is a rigid structure, the vehicle tire slip angle tends to be small under normal driving conditions. In order to facilitate the study of vehicle motion, the vehicle dynamics bicycle model is selected as:

$$m(\dot{v}_x - v_y \dot{\phi}) = F_{fy} \sin \delta_f + F_{fx} \cos \delta_f + F_{rx}, \quad (2.1)$$

$$m(\ddot{y} + v_x \dot{\varphi}) = F_{fy} \cos \delta_f + F_{fx} \sin \delta_f + F_{ry} - F_{af}, \quad (2.2)$$

$$F_{af} = F_a + F_f. \quad (2.3)$$

where m is mass, φ is yaw angle. Lateral and longitudinal velocity are v_x, v_y . F_{iy} ($I=f, r$) and F_{ix} ($I=f, r$) are the lateral and longitudinal tire forces. δ_f is front wheel steering angle, the air resistance F_a is proportional to V_l , F_f is the rolling resistance, they can be expressed as:

$$F_a = \frac{1}{2} C \rho A V_l^2, \quad (2.4)$$

$$F_f = (\mu_0 + \mu_1 v_{wx}^2) mg, \quad (2.5)$$

where ρ , C , and A are the density parameter, air resistance coefficient and windward area. μ_0 , μ_1 are nominal and variable value of resistance coefficient respectively, and v_{wx} is the longitudinal velocity of the tire.

$$I_z \ddot{\varphi} = l_f (F_{fy} \cos \delta_f + F_{fx} \sin \delta_f) - l_r F_{ry}, \quad (2.6)$$

where I_z is moment of inertia, l_f and l_r are the distance from the front and rear axles to the center of vehicle. The longitudinal driving force F_y of the tire can be expressed as:

$$F_y = f_y(\alpha, F_z, s, \mu). \quad (2.7)$$

Tire slip angle α is calculated by v_{wx} and v_{wy} :

$$\alpha = \tan^{-1} \left(\frac{v_{wx}}{v_{wy}} \right), \quad (2.8)$$

where v_{wx} is the longitudinal velocity of the tire, v_{wy} is the lateral velocity of the tire. They can be represented:

$$\begin{cases} v_{wfx} = v_x \cos \delta + (v_y + a\dot{\varphi}) \sin \delta \\ v_{wrx} = v_x \cos \delta + (v_y - b\dot{\varphi}) \sin \delta \end{cases} \quad (2.9)$$

$$\begin{cases} v_{wfy} = v_x \cos \delta - (v_y + a\dot{\varphi}) \sin \delta \\ v_{wry} = v_x \cos \delta - (v_y - b\dot{\varphi}) \sin \delta \end{cases} \quad (2.10)$$

Due to the difference between wheel speed and actual speed during the actual vehicle movement, the desired slip rate s_r is as follows:

$$s_r = \begin{cases} 1 - \frac{v_{wx}}{rw_w} (rw_w \neq 0, rw_w > v_{wx}) \\ \frac{rw_w}{v_{wx}} - 1 (v_{wx} \neq 0, rw_w < v_{wx}) \end{cases} \quad (2.11)$$

where r and w_w are the radius and angular velocity of the tire.

The vertical load on the front and rear tires of the vehicle can be expressed as the following equation:

$$\begin{cases} F_{zfl, zfr} = \frac{l_r mg}{2(l_r + l_f)} \mp \Delta F_1 + \Delta F_2 \\ F_{zrl, zrr} = \frac{l_f mg}{2(l_r + l_f)} + \Delta F_1 \mp \Delta F_2 \end{cases} \quad (2.12)$$

where ΔF_1 , ΔF_2 denote the longitudinal and lateral transfer loads of vehicles, respectively, and the specific equations are as follows:

$$\begin{cases} \Delta F_1 = \frac{m(\dot{v}_x - v_y w) h_{cog}}{2(l_r + l_f)} \\ \Delta F_2 = \frac{m(\dot{v}_x - v_y w) h_{cog}}{2L_h} + \frac{gm_s h_s}{2L_h} \left(\frac{h_{cog} \xi}{g} - \sin \xi \right) \end{cases} \quad (2.13)$$

where h_{cog} is the height between the center of mass and the ground, L_h is half of wheel base, and h_s is the suspension height of vehicle body. ξ is slip angle of wheel.

Based on the assumption that the tire slip angle tends to be small under normal driving conditions, the tire slip force F_{fy} , F_{ry} are simplified as:

$$\begin{cases} F_{fy} = N_{af} \alpha_f \\ F_{ry} = N_{ar} \alpha_r \end{cases} \quad (2.14)$$

where N_{af} and N_{ar} are the cornering stiffness. From vehicle kinematics model, it can be gained that the front wheel angle α_f and rear wheel angle α_r :

$$\begin{cases} \alpha_f = \delta_f - \frac{v_y + l_f \dot{\phi}}{v_x} \\ \alpha_r = -\frac{v_y + l_r \dot{\phi}}{v_x} \end{cases} \quad (2.15)$$

Considering the small front wheel angle, $\cos \delta \approx 1$ and $\sin \delta \approx 0$. At the same time, we suppose the AGEV only moves in a plane coordinate system, and the aerodynamics and rolling effect of the motion process is not considered in order to study. Therefore, according to Eqs (2.14) and (2.15), the Eqs (2.2) and (2.6) are rewritten as:

$$\ddot{y} = -\frac{N_{\alpha_f} + N_{\alpha_f}}{mv_x} \dot{y} - \left(\frac{l_f N_{\alpha_f} - l_r N_{\alpha_f}}{mv_x} + v_x \right) \dot{\phi} + w_2, \quad (2.16)$$

$$\ddot{\phi} = -\frac{l_f N_{\alpha_f} - l_r N_{\alpha_f}}{I_z v_x} \dot{y} - \frac{l_f^2 N_{\alpha_f} + l_r^2 N_{\alpha_f}}{I_z v_x} \dot{\phi} + w_4 \quad (2.17)$$

where w_2 and w_4 represent the error of the lateral and yaw angular acceleration.

2.2. Vehicle trajectory following formulation

When the AGEV is following the desired trajectory, the current and expected position and heading information of the AGEV are what we need to care about. Figure 1 shows diagram of AGEV trajectory following process, ϕ , ϕ_r are the current yaw angles and expected yaw angles, and the first derivatives of lateral error y_e and yaw angle error ϕ_e can be written as:

$$\dot{y}_e = v_y - v_x \phi_e, \quad (2.18)$$

$$\dot{\phi}_e = \dot{\phi} - \dot{\phi}_r = \dot{\phi} - \chi \dot{s} \quad (2.19)$$

where s , χ are the actual trajectory and the curvature of the road. The first and second derivatives of s can be described as:

$$\dot{s} = v_x + v_y \phi_e, \quad (2.20)$$

$$\ddot{s} = \dot{v}_x + \dot{v}_y \phi_e + v_y \dot{\phi}_e. \quad (2.21)$$

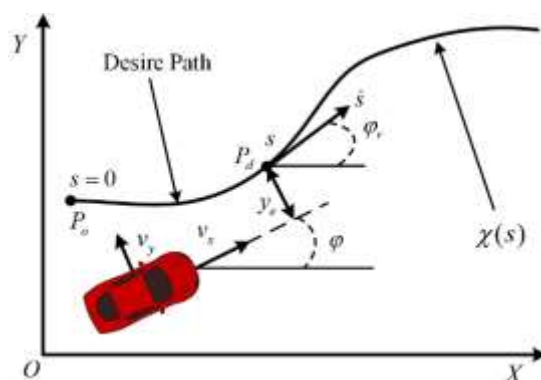


Figure 1. Vehicle trajectory following process.

From the Eqs (2.18) to (2.21), and the second derivatives of y_e and ϕ_e can be converted to the Eqs (2.22) and (2.23).

$$\ddot{y}_e = \dot{v}_y - \dot{v}_x \varphi_e - v_x \dot{\varphi}_e, \quad (2.22)$$

$$\ddot{\varphi}_e = \dot{\zeta} - \dot{\chi} \dot{s} - \chi \ddot{s}. \quad (2.23)$$

2.3. Uncertain vehicle trajectory following model

According to a series of formula derivations and equation transformations, the above vehicle dynamics equation is rewritten as a state space expression:

$$\dot{x} = Ax + B_u u + B_w w \quad (2.24)$$

where w , B_w are model disturbances and coefficient matrix to state space, $u = \delta_f$ is the control input. The expressions for system matrices x , A , B_u , w and B_w are as follows:

$$x = \begin{bmatrix} y_e \\ \dot{y}_e \\ \varphi_e \\ \dot{\varphi}_e \end{bmatrix}, A = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & -\frac{N_{\alpha f} + N_{\alpha r}}{mv_x} & \frac{N_{\alpha f} + N_{\alpha r}}{m} & -\frac{l_f N_{\alpha f} - l_r N_{\alpha r}}{mv_x} \\ 0 & 0 & 0 & 1 \\ 0 & -\frac{l_f N_{\alpha f} - l_r N_{\alpha r}}{I_z v_x} & \frac{l_f N_{\alpha f} - l_r N_{\alpha r}}{I_z} & -\frac{l_f^2 N_{\alpha f} + l_r^2 N_{\alpha r}}{I_z v_x} \end{bmatrix}$$

$$B_u = \begin{bmatrix} 0 & \frac{N_{\alpha f}}{m} & 0 & \frac{l_f N_{\alpha f}}{I_z} \end{bmatrix}^T, B_w = [0 \ 1 \ 0 \ 1]$$

The relationship between the steering wheel angle and the front wheel angle of the steering system can be described by the following formula:

$$\delta_a = \tau \delta_f \quad (2.25)$$

where τ is the transmission ratio of the steering system.

When vehicles travel on complex and changeable roads, due to changes in steering angles and road surface roughness, the value of tire cornering stiffness N_i ($i = \alpha f, \alpha r$) is constantly changing and bounded. It can be handled as follows:

$$\begin{cases} N_{\alpha f} = \bar{N}_{\alpha f} + n_{\alpha f} \tilde{N}_{\alpha f}, \\ N_{\alpha r} = \bar{N}_{\alpha r} + n_{\alpha r} \tilde{N}_{\alpha r} \end{cases}, |n_i| < 1 (i = \alpha f, \alpha r), \quad (2.26)$$

$$\bar{N}_{\alpha j} = \frac{N_{\alpha j \min} + N_{\alpha j \max}}{2} (j = f, r), \quad (2.27)$$

$$\tilde{N}_{\alpha j} = \frac{N_{\alpha j \max} - N_{\alpha j \min}}{2} \quad (j = f, r) \quad (2.28)$$

where $N_{i \max}$, $N_{i \min}$ are the maximal and minimal values of N_i ($i = \alpha f, \alpha r$). The system's n_i time varying parameters meet $|n_i| \leq 1$ ($i = \alpha f, \alpha r$).

It is possible to change the vehicle system model (2.24) of the as follows:

$$\dot{x} = A_d x + B_{ud} u + B_w w \quad (2.29)$$

where the arguments have the following meaning:

$$A_d = \bar{A}_d + \Delta A, B_{ud} = \bar{B}_{ud} + \Delta B_u, [\Delta A \quad \Delta B_u] = H_d F_d [E_A \quad E_B], |F_d| \leq 1,$$

$$\bar{A}_d = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & -\frac{\bar{N}_{\alpha f} + \bar{N}_{\alpha r}}{m v_x} & \frac{\bar{N}_{\alpha f} + \bar{N}_{\alpha r}}{m} & -\frac{l_f \bar{N}_{\alpha f} - l_r \bar{N}_{\alpha r}}{I_z} \\ 0 & 0 & 0 & 1 \\ 0 & -\frac{l_f \bar{N}_{\alpha f} - l_r \bar{N}_{\alpha r}}{I_z v_x} & \frac{l_f \bar{N}_{\alpha f} - l_r \bar{N}_{\alpha r}}{I_z} & -\frac{l_f^2 \bar{N}_{\alpha f} + l_r^2 \bar{N}_{\alpha r}}{I_z v_x} \end{bmatrix}, \bar{B}_{ud} = \begin{bmatrix} 0 \\ \frac{\bar{N}_{\alpha f}}{m} \\ 0 \\ \frac{l_f \bar{N}_{\alpha f}}{I_z} \end{bmatrix},$$

$$E_A = \begin{bmatrix} 0 & -\frac{1}{m v_x} & \frac{1}{m} & -\frac{l_f}{m v_x} \\ 0 & -\frac{1}{m v_x} & \frac{1}{m} & \frac{l_r}{m v_x} \\ 0 & -\frac{l_f}{I_z v_x} & \frac{l_f}{I_z} & -\frac{l_f^2}{I_z v_x} \\ 0 & \frac{l_r}{I_z v_x} & -\frac{l_r}{I_z} & -\frac{l_r^2}{I_z v_x} \end{bmatrix}, H_d = \begin{bmatrix} 0 & 0 & 0 & 0 \\ \tilde{N}_{\alpha f} & \tilde{N}_{\alpha r} & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & \tilde{N}_{\alpha f} & \tilde{N}_{\alpha r} \end{bmatrix}, E_B = \begin{bmatrix} \frac{1}{m} \\ 0 \\ \frac{l_f}{I_z} \\ 0 \end{bmatrix}.$$

3. Nonlinear robust controller design

3.1. Design of robust feedback control

The trajectory-following system takes into account a number of objectives, including as lateral error, yaw error and vehicle workload. In order to track the target, we define the error function and design of robust linear feedback gain as:

$$J_s = \int_0^{\infty} q_1 y_e^2 + q_2 \dot{y}_e^2 + q_3 \varphi_e^2 + q_4 \dot{\varphi}_e^2 + q_4 \delta^2 dt \quad (3.1)$$

where the meaning of the letter symbol is as follows:

$$y_e = y_a - y_d, \dot{y}_e = v_y - v_x \varphi_e, \varphi_e = \varphi - \varphi_r, \dot{\varphi}_e = \dot{\varphi} - \chi \dot{s}.$$

The above equation can be written as:

$$\begin{aligned} J &= \int_0^\infty [(\bar{C}_1 x + \bar{D}_{11} w)^T U (\bar{C}_1 x + \bar{D}_{11} w) + u^T V u] dt \\ &= \int_0^\infty [(U^{\frac{1}{2}} \bar{C}_1 x + U^{\frac{1}{2}} \bar{D}_{11} w)^T (U^{\frac{1}{2}} \bar{C}_1 x + U^{\frac{1}{2}} \bar{D}_{11} w) + \int_0^\infty (V^{\frac{1}{2}} u)^T (V^{\frac{1}{2}} u) dt \end{aligned} \quad (3.2)$$

where the arguments have the following meaning:

$$\bar{C}_1 = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}, \bar{D}_{11} = 1, U = \begin{bmatrix} q_1 & 0 & 0 & 0 \\ 0 & q_2 & 0 & 0 \\ 0 & 0 & q_3 & 0 \\ 0 & 0 & 0 & q_4 \end{bmatrix}, V = q_5$$

The control output of the model can be defined as:

$$z = C_1 x + D_{11} w + D_{12} u \quad (3.3)$$

where

$$C_1 = \begin{bmatrix} U^{\frac{1}{2}} \bar{C}_1 \\ 0 \end{bmatrix}, D_{11} = \begin{bmatrix} U^{\frac{1}{2}} \bar{D}_{11} \\ 0 \end{bmatrix}, D_{12} = \begin{bmatrix} 0 \\ V^{\frac{1}{2}} \end{bmatrix}.$$

The optimization of the error performance function J can be translated into the optimization of the control output z in order to facilitate the design of the trajectory tracking controller. The error cost function J has the following relationship with the control output z :

$$J = \|z\|_2^2. \quad (3.4)$$

Based on the above system model of vehicle dynamics (2.29) and the control output of the system (3.3), vehicle trajectory following with parameter uncertainties can be transformed into a standard H_∞ -infinity control problem.

$$\begin{cases} \dot{x} = A_d x + B_w w + B_{ud} u \\ z = C_1 x + D_{11} w + D_{12} u \end{cases} \quad (3.5)$$

According to H_∞ control theory, the purpose of this system model is to design a control gain K and state feedback controller $u_L = Kx$ to meet the vehicle trajectory following requirements.

From Eq (3.5) and $u_L = Kx$, the following closed-loop system of vehicle trajectory following can be obtained.

$$\begin{cases} \dot{x} = A_s x + B_s w \\ z = C_s x + D_s w \end{cases} \quad (3.6)$$

where:

$$A_s = \bar{A}_d + \bar{B}_{ud}K + H_d F_d (E_A + E_B K), \quad B_s = B_w, \quad C_s = C_1 + D_{12}K, \quad D_s = D_{11}.$$

The sufficient and necessary conditions for vehicle trajectory tracking system to be stable are that A_s and B_s are stable, and A_s and C_s are detectable. In this paper, we regard the system model error w as the external perturbation of the system. Considering the system (3.6), its transfer function is:

$$T(s) = C_s (sI - A_s) B_s + D_s. \quad (3.7)$$

The input signal $w(t)$ and output signal energy $z(t)$ are respectively defined as:

$$\|\Xi\|_2^2 = \int_0^{+\infty} \Xi^T(t) \Xi(t) dt, \quad \Xi = w, z. \quad (3.8)$$

The H_∞ norm of $T(s)$ is defined as:

$$\|T(s)\|_\infty = \sup_{w \neq 0} \frac{\|z\|_2}{\|w\|_2}. \quad (3.9)$$

The equality (3.9) can be changed to:

$$\|T(s)\|_\infty = \sup_w \sqrt{\frac{E(z^T(t)z(t))}{E(w^T(t)w(t))}} \quad (3.10)$$

where $E(\cdot)$ is mathematical expectation.

That is, the peak value of the maximum singular value of the system's frequency response. In fact, the design of H-infinity state feedback controller is to find the optimal K meet the performance of tracking system, and the influence of perturbation on the output performance is controlled under a certain level. Therefore, the following H_∞ performance index is selected:

$$\int_0^\infty z^T(t)z(t)dt < \gamma^2 \int_0^\infty w^T(t)w(t)dt. \quad (3.11)$$

The above inequality has the following equivalence relation.

$$\|G(s)\|_\infty < \gamma \Leftrightarrow \sigma_{\max}[G(jw)] < \gamma \Leftrightarrow \gamma^2 I > G(jw)^T G(jw). \quad (3.12)$$

To prove the closed-loop system of path following (3.6) is stable and meets H_∞ performance, some lemmas will be introduced.

Lemma 1. Given matrix $L = \begin{bmatrix} L_{11} & L_{12} \\ L_{21} & L_{22} \end{bmatrix}$, the dimension of L_{11} is $r \times r$, where $L = L^T$, the following three equivalent conditions can be obtained:

$$L < 0, \quad (3.13)$$

$$L_{11} < 0, L_{22} - L_{12}^T L_{11}^{-1} L_{12} < 0, \quad (3.14)$$

$$L_{22} < 0, L_{11} - L_{12}L_{22}^{-1}L_{12}^T < 0. \quad (3.15)$$

Lemma 2. Given an appropriate dimension matrix $Q = Q^T$, B and J are symmetric matrices, B and J are real matrices, and $F^T F \leq 1$. Then there is the following inequation:

$$Q + BFJ + J^T F^T B^T < 0 \quad . \quad (3.16)$$

The sufficient and necessary condition is that there exists a positive value $\varepsilon > 0$ satisfy following inequation:

$$Q + \varepsilon BB^T + \varepsilon^{-1} J^T J < 0. \quad (3.17)$$

For the above closed-loop system (3.6) for vehicle path tracking, the H_∞ control gain K can be solved by the developed Theorem 1.

Theorem 1. The stability of path following system is considered, the conditions for a closed-loop system (3.6) to satisfy the H_∞ performance are that there are symmetric matrices $X > 0$, $V > 0$, and positive numbers γ , ε , satisfy the following inequation (3.18), and there exists a control gain K of H_∞ in the system such that:

$$\begin{bmatrix} \text{syms}(\bar{A}_d X + \bar{B}_{ud} V) & B_w & \Upsilon_1 & \varepsilon H_d & \Upsilon_2 \\ * & -\gamma I & D_{11}^T & 0 & 0 \\ * & * & -\gamma I & 0 & 0 \\ * & * & * & -\varepsilon I & 0 \\ * & * & * & * & -\varepsilon I \end{bmatrix} < 0 \quad (3.18)$$

where

$$\text{syms}(\ast) = \ast + \ast^T, \Upsilon_1 = XC_1^T + V^T D_{12}^T, \Upsilon_2 = XE_A^T + V^T E_B^T.$$

Proof. According to Lyapunov stability theory. For the closed-loop system (3.6), if there is a symmetric positive definite matrix P satisfies that the inequality (3.19) holds for all uncertain parameters $\delta \in \Delta$.

$$A_s^T(\delta)P + PA_s(\delta) < 0. \quad (3.19)$$

That is, there exist appropriate positive definite n -order matrices $P > 0$ and scalars $\beta > 0$ such that:

$$x^T(t)[A_s^T P + PA_s]x(t) + 2x(t)\Delta A_s^T(t)Px(t) \leq -\beta \|x(t)\|^2. \quad (3.20)$$

Define the Lyapunov function:

$$V(x(t)) = x(t)^T Px(t). \quad (3.21)$$

And assuming that:

$$\begin{cases} e_1 = \sigma_{\min}(P) \\ e_2 = \sigma_{\max}(P) \end{cases}. \quad (3.22)$$

There are:

$$e_1 \|x(t)\|^2 \leq V(x(t)) \leq e_2 \|x(t)\|^2, \quad (3.23)$$

$$\frac{dV(x(t))}{dt} = x^T(t)[PA_s + A_s^T P]x(t) + 2x(t)\Delta A_s^T(t)Px(t) \leq -\alpha \|x(t)\|^2. \quad (3.24)$$

Stability of closed-loop systems means that for any $\Delta A(t)$ is exponentially stable, and for any initial state $x(0)$, there is:

$$\|x(t)\|^2 \leq \frac{\sigma_{\max}(P)}{\sigma_{\min}(P)} \|x(0)\|^2 \exp\left[-\frac{\beta}{\sigma_{\min}(P)} t\right]. \quad (3.25)$$

According to the Raccati theory, there being a positive definite matrix $P > 0$ such that:

$$PA_0 + A_0^T P + PH_d H_d^T P + E^T E < 0. \quad (3.26)$$

Among them:

$$A_0 = \bar{A}_d + \bar{B}_{ud} K, E = E_A + E_B K. \quad (3.27)$$

Assume:

$$Q = -(A_0^T P + PA_0 + PH_d H_d^T P + E^T E) > 0. \quad (3.28)$$

Define $V(x) = x^T P x$, then there is the following relationship:

$$\begin{aligned} \frac{dV(x(t))}{dt} &= x^T (A_0^T P + PA_0)x + 2x\Delta A_0^T P x \\ &= x^T (-Q - PH_d H_d^T P - E^T E)x + 2x^T E^T F_d^T E^T P x \\ &\leq -x^T Q x - x^T PH_d H_d^T P x - x^T E^T F_d^T F_d E x + 2xE^T F_d^T E^T P x \\ &= -x^T Q x - x^T (PH_d - E^T F_d^T)(H_d^T P - F_d E)x \\ &\leq -\beta \|x\|^2 \end{aligned} \quad (3.29)$$

where $\alpha > 0$ satisfies:

$$\alpha \leq \sigma_{\min}(Q). \quad (3.30)$$

Thus, sufficiency of the closed-loop system (3.6) stability is proved.

Next, it is proved whether the concerned H_∞ performance condition is satisfied in the presence of road disturbance.

The conditions for the system (3.6) to be asymptotically and satisfies H_∞ performance condition (3.11) is that there exists a symmetric positive definite matrix $P > 0$ and scale γ that satisfies the following matrix inequation:

$$\begin{bmatrix} PA_s + A_s^T P & PB_s & C_s^T \\ * & -\gamma I & D_s^T \\ * & * & -\gamma I \end{bmatrix} < 0. \quad (3.31)$$

According to closed-loop system (3.6), inequation (3.31) can be written as the following conditions:

$$\begin{bmatrix} \text{sym}(P(\bar{A}_d + \bar{B}_{ud}K) + PH_d F_d(E_A + E_B K)) & PB_w & (C_1 + D_{12}K)^T \\ * & -\gamma I & D_{11}T \\ * & * & -\gamma I \end{bmatrix} < 0. \quad (3.32)$$

Inequation (3.32) can be written as the following conditions.

$$\begin{bmatrix} \text{sym}(P(\bar{A}_d + \bar{B}_{ud}K)) & PB_w & C_s^T \\ * & -\gamma I & D_{11}^T \\ * & * & -\gamma I \end{bmatrix} + \begin{bmatrix} \text{sym}(PH_d F_d(E_A + E_B K)) & 0 & 0 \\ * & 0 & 0 \\ * & * & 0 \end{bmatrix} < 0. \quad (3.33)$$

Inequation (3.33) is equivalent to the following:

$$\begin{bmatrix} \text{sym}(P(\bar{A}_d + \bar{B}_{ud}K)) & PB_w & C_s^T \\ * & -\gamma I & D_{11}^T \\ * & * & -\gamma I \end{bmatrix} + \begin{bmatrix} PH_d \\ 0 \\ 0 \end{bmatrix} F_d [(E_A + E_B K) \quad 0 \quad 0] \\ + \begin{bmatrix} (E_A + E_B K) \\ 0 \\ 0 \end{bmatrix} F_d^T [PH_d \quad 0 \quad 0] < 0. \quad (3.34)$$

Let

$$\bar{H}_d = [PH_d \quad 0 \quad 0]^T, \bar{F}_d = F_d, \bar{E}_{AB} = [(E_A + E_B K) \quad 0 \quad 0], \varpi = \begin{bmatrix} \Gamma_1 & PB_w & C_{cl}^T \\ * & -\gamma I & D_{11}T \\ * & * & -\gamma I \end{bmatrix}.$$

Inequation (3.34) can be written as the following conditions:

$$\varpi + \bar{H}_d \bar{F}_d \bar{E}_{AB} + \bar{E}_{AB}^T \bar{F}_d^T \bar{H}_d < 0. \quad (3.35)$$

According to Lemma 2, there exists a positive value $\varepsilon_1 > 0$ satisfy following inequation:

$$\varpi + \varepsilon_1 \bar{H}_d \bar{H}_d^T + \varepsilon_1^{-1} \bar{E}_{AB} \bar{E}_{AB}^T < 0. \quad (3.36)$$

According to Lemma 1, the following inequation can be gained.

$$\begin{bmatrix} \varpi & \bar{H}_d & \bar{E}_{AB}^T \\ & -\varepsilon_1^{-1}I & 0 \\ & & \varepsilon_1 I \end{bmatrix} < 0. \quad (3.37)$$

Expand the above inequality (3.37):

$$\begin{bmatrix} \text{syms}(P(\bar{A}_d + \bar{B}_{ud}K)) & PB_w & (C_1 + D_{12}K)^T & PH_d & (E_A + E_B K)^T \\ * & -\gamma I & D_{11}^T & 0 & 0 \\ * & * & -\gamma I & 0 & 0 \\ * & * & * & \varepsilon_1^{-1}I & 0 \\ * & * & * & * & \varepsilon_1 I \end{bmatrix} < 0. \quad (3.38)$$

The diagonal matrix $\text{diag}\{P^{-1}, I, I, \varepsilon_1, I\}$ is multiplied to the left and right of the above formula respectively, and the following inequation (3.39) is obtained.

$$\begin{bmatrix} \text{syms}((\bar{A}_d + \bar{B}_{ud}K)P^{-1}) & B_w & P^{-1}(C_1 + D_{12}K)^T & \varepsilon_1 H_d & P^{-1}(E_A + E_B K)^T \\ * & -\gamma I & D_{11}^T & 0 & 0 \\ * & * & -\gamma I & 0 & 0 \\ * & * & * & \varepsilon_1 I & 0 \\ * & * & * & * & \varepsilon_1 I \end{bmatrix} < 0. \quad (3.39)$$

Let $X = P^{-1}$, the following inequation (3.40) is obtained.

$$\begin{bmatrix} \text{syms}((\bar{A}_d + \bar{B}_{ud}K)X) & B_w & X(C_1 + D_{12}K)^T & \varepsilon_1 H_d & X(E_A + E_B K)^T \\ * & -\gamma I & D_{11}^T & 0 & 0 \\ * & * & -\gamma I & 0 & 0 \\ * & * & * & \varepsilon_1 I & 0 \\ * & * & * & * & \varepsilon_1 I \end{bmatrix} < 0. \quad (3.40)$$

Then let $KX = V$, by the fundamental transformation of the inequality, the Theorem 1 can be obtained.

3.2. Nonlinear robust control feedback design

Next, so as to improve the system's fast response, transient response performance and reduce overshoot, the nonlinear compensation feedback control rate will be designed as:

$$u_{\text{nonlinear}} = \varphi(r, h)B^T Px \quad (3.41)$$

where $\varphi(r, h)$ is nonlinear compensation function, x is error state, r, h are output reference value and actual value respectively. The matrix P is the only positive definite matrix of the following Lyapunov equation, and the asymptotic stability of A_s also guarantees the existence of matrix P .

$$A_s^T P + P A_s + W = 0 \quad (3.42)$$

where $W \in R^{n \times n}$ is positive definite weighted diagonal matrix. The selection of W can determine stability closed-loop poles of the system. The principle of matrix selection is to make the closed-loop pole have greater damping when the system reaches the set goal. Thus, the system overshoot near the given reference signal is suppressed, the adjustment time of the system is shortened, and the transient performance of the system is improved.

$$W = 10^g \cdot I \quad (3.43)$$

where g is an adjustable parameter, I is the corresponding dimension of the unit matrix.

The selection of nonlinear function $\varphi(r, h)$ is very crucial, and its selection needs to satisfy the following two basic properties: (1) In the early stage of control, when the error is relatively large, the value of the nonlinear function $\varphi(r, h)$ needs to become smaller, so the nonlinear feedback compensation has no obvious effect on the system. At this time, the linear feedback controller plays a leading role and gives priority to meeting the needs of the control system's rapidity. (2) When the control is close to the steady state, the error is relatively small, and the value of the nonlinear function should become larger and occupy a dominant position. By continuously increasing the system damping, it will meet demand for fast response speed and low overshoot.

From the above analysis, the form of nonlinear compensation function selected in this paper is as follows:

$$\varphi(r, h) = -\beta \frac{1}{1 - e^{-1}} \left(e^{-\alpha \left| \frac{h-h_0}{r-h_0} \right|} - e^{-1} \right) \quad (3.44)$$

where r and h are reference outputs and control outputs respectively, h_0 is the initial value of the control output. α and β are non-negative adjustable parameters, e is natural number.

Combining the linear feedback and the nonlinear compensation function, the output of the actuator is finally formed in the following form (3.45). The linear gain feedback can bring a faster response to the trajectory following system, at the same time, the nonlinear compensation function can obtain a stable output and reduce the overshoot of the system.

$$u_{final} = Kx + \varphi(r, h) B_{ud}^T P x. \quad (3.45)$$

According to the design of the above nonlinear compensation function and considering that the output of the actuator of the system will be saturated, the following form of nonlinear robust control system model can be rewritten:

$$\begin{cases} \dot{x} = A_d x + B_w w + B_{ud} \text{sat}(u_{final}) \\ z = C_1 x + D_{11} w + D_{12} \text{sat}(u_{final}) \end{cases} \quad (3.46)$$

where $\text{sat}(\ast)$ represents the saturation function of the system, $\text{sat}(u_{final})$ can be expressed as:

$$\text{sat}(u_{final}) = \begin{cases} u_{max}, & u_{final} > u_{max} \\ Kx + \varphi(r, h)B_{ud}^T Px, & |u_{final}| < u_{max} \\ -u_{max}, & u_{final} < -u_{max} \end{cases} \quad (3.47)$$

Considering that the front wheel angle will be saturated, the actual nonlinear compensation can be expressed as:

$$u_{nla} = \text{sat}(u) - Kx. \quad (3.48)$$

According to the above conditions, u_{nla} can be written as

$$0 < |u_{nla}| < \varphi(r, h)B_{ud}^T Px. \quad (3.49)$$

Next, the influence of nonlinear compensation function on the stability and H-infinity performance of the system will be proved.

Proof. The Lyapunov functional is first defined as

$$V = x^T Px. \quad (3.50)$$

Take the derivative of the Lyapunov functional:

$$\begin{aligned} \dot{V} &= \dot{x}^T Px + x^T P\dot{x} \\ &= (\bar{A}x + \bar{R}x + B_w w)^T Px + x^T P(\bar{A}x + \bar{R}x + B_w w) \\ &= x^T \bar{A}^T Px + x^T \bar{R}^T Px + w^T B_w^T Px + x^T P\bar{A}x + x^T P\bar{R}x + x^T PB_w w \\ &= x^T (\bar{A}^T P + P\bar{A})x + x^T (\bar{R}^T P + P\bar{R})x + w^T B_w^T Px + x^T PB_w w \end{aligned} \quad (3.51)$$

where

$$\bar{A} = A_d + B_{ud}K, \bar{R} = B_{ud}u_{nla}.$$

When there is no disturbance outside the system, it is assumed that $w = 0$, and the Lyapunov functional can be expressed as:

$$\dot{V} = x^T (\bar{A}^T P + P\bar{A})x + x^T (\bar{R}^T P + P\bar{R})x. \quad (3.52)$$

Assuming that

$$V_1 = x^T (\bar{A}^T P + P\bar{A})x. \quad (3.53)$$

It is known from the matrix inequality that V_1 is less than zero.

Assuming that

$$V_2 = x^T (\bar{R}^T P + P \bar{R}) x = 2x^T P B_{ud} (\text{sat}(Kx + \varphi B_{ud}^T P x) - Kx) = 2i(\text{sat}(j + \varphi i) - j) \quad (3.54)$$

where $i = x^T P B_{ud}$, $j = Kx$.

V_2 will be discussed in two cases.

(1) When the input is not saturated, it satisfies:

$$|\text{sat}(j + \varphi i)| < u_{\max}. \quad (3.55)$$

Then there is the following inequality:

$$V_2 = 2i(\text{sat}(j + \varphi i) - j) = 2\varphi i^2 \leq 0. \quad (3.56)$$

Therefore:

$$\dot{V} = V_1 + V_2 = x^T (\bar{A}^T P + P \bar{A}) x + 2i(\text{sat}(j + \varphi i) - j) \leq x^T (\bar{A}^T P + P \bar{A}) x < 0. \quad (3.57)$$

(2) When the input is saturated, it means:

$$|\text{sat}(j + \varphi i)| \geq u_{\max}. \quad (3.58)$$

Consider that when j is also saturated, when $\varphi i = 0$, there is only linear feedback at this time, and the system is asymptotically stable.

When j is unsaturated, the output has the following two forms:

$$\begin{cases} i < 0, \text{sat}(j + \varphi i) - j \geq 0, & \text{When } j + \varphi i \geq u_{\max} \text{ and } j > 0 \\ i > 0, \text{sat}(j + \varphi i) - j \leq 0, & \text{When } j + \varphi i \leq -u_{\max} \text{ and } j < 0 \end{cases}. \quad (3.59)$$

It can be seen from the condition of inequality (3.59) that when saturation occurs, the signs of i and $\text{sat}(j + \varphi i) - j$ are opposite, that is:

$$V_2 = 2i(\text{sat}(j + \varphi i) - j) \leq 0. \quad (3.60)$$

Thus:

$$\dot{V} = V_1 + V_2 = x^T (\bar{A}^T P + P \bar{A}) x + 2i(\text{sat}(j + \varphi i) - j) \leq x^T (\bar{A}^T P + P \bar{A}) x < 0. \quad (3.61)$$

Therefore, the system is also closed-loop stable with the introduction of nonlinear compensation function and without external interference.

Next, it will be proved that the system with nonlinear compensation function is closed-loop stable when subjected to external interference and satisfies the H_∞ performance index. First let's define a cost function J_c :

$$J_c = \dot{V} + z^T z - \gamma^2 w^T w. \quad (3.62)$$

Since the system is asymptotically stable, then if the H_∞ norm has the following inequality:

$$\|z\|^2 < \gamma^2 \|w\|^2. \quad (3.63)$$

In other words, the following inequality exists:

$$J_c = \dot{V} + z^T z - \gamma^2 w^T w < 0. \quad (3.64)$$

According to the above conditions:

$$\begin{aligned} \dot{V} &= \dot{x}^T P x + x^T P \dot{x} \\ &= x^T \bar{A}^T P x + x^T \bar{R}^T P x + w^T B_w^T P x + x^T P \bar{A} x + x^T P \bar{R} x + x^T P B_w w. \\ &= x^T (\bar{A}^T P + P \bar{A} + 2P \bar{R}) x + w^T B_w^T P x + x^T P B_w w \end{aligned} \quad (3.65)$$

It could also be written:

$$\dot{V} = \begin{bmatrix} x \\ w \end{bmatrix}^T \begin{bmatrix} \bar{A}^T P + P \bar{A} + 2P \bar{R} & P B_w \\ B_w^T P & 0 \end{bmatrix} \begin{bmatrix} x \\ w \end{bmatrix}. \quad (3.66)$$

In the same way:

$$z^T z = (Cx + Dw)^T (Cx + Dw) = x^T C^T C x + x^T C^T D w + w^T D^T C x D w. \quad (3.67)$$

The above equation can be further rewritten:

$$z^T z = \begin{bmatrix} x \\ w \end{bmatrix}^T \begin{bmatrix} C^T C & C^T D \\ D^T C & D^T D \end{bmatrix} \begin{bmatrix} x \\ w \end{bmatrix}, \quad (3.68)$$

$$J_c = \begin{bmatrix} x \\ w \end{bmatrix}^T \left(\begin{bmatrix} \bar{A}^T P + P \bar{A} + 2P \bar{R} & P B_w \\ B_w^T P & 0 \end{bmatrix} + \begin{bmatrix} C^T C & C^T D \\ D^T C & D^T D \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & \gamma^2 \end{bmatrix} \right) \begin{bmatrix} x \\ w \end{bmatrix}. \quad (3.69)$$

According to the properties of quadratic form and inequality (3.62), the following inequality holds:

$$\Psi = \begin{bmatrix} \bar{A}^T P + P \bar{A} + 2P \bar{R} & P B_w \\ B_w^T P & 0 \end{bmatrix} + \begin{bmatrix} C^T C & C^T D \\ D^T C & D^T D \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & \gamma^2 \end{bmatrix} < 0. \quad (3.70)$$

Rewrite the above inequation:

$$\Psi = \begin{bmatrix} \bar{A}^T P + P \bar{A} + 2P \bar{R} & P B_w \\ B_w^T P & 0 \end{bmatrix} + \begin{bmatrix} C^T C & C^T D \\ D^T C & D^T D \end{bmatrix} = \begin{bmatrix} \bar{A}^T P + P \bar{A} + 2P \bar{R} & P B_w \\ B_w^T P & \gamma^2 I \end{bmatrix} + \begin{bmatrix} C^T \\ D^T \end{bmatrix} [C \ D]. \quad (3.71)$$

Let $\pi = [C \ D]$, the above inequality has the following relation:

$$\Psi = \begin{bmatrix} \bar{A}^T P + P \bar{A} + 2P \bar{R} & P B_w \\ B_w^T P & \gamma^2 I \end{bmatrix} + \pi^T \pi. \quad (3.72)$$

According to Lemma 1:

$$\Psi = \begin{bmatrix} \bar{A}^T P + P\bar{A} + 2P\bar{R} & PB_w & C^T \\ B_w^T P & \gamma^2 I & D^T \\ C & D & -I \end{bmatrix}. \quad (3.73)$$

According to Theorem 1 and inequalities (3.54), (3.56) and (3.60), the above equation is less than 0. Thus, the theorem of robust state feedback control with nonlinear compensation function is proved.

Remark 1. Note that, in this study, the nonlinear robust control proposed in this paper includes linear feedback gain and nonlinear feedback gain parts, which can combine the advantages of linear feedback control and nonlinear feedback control. Since the direct design of fully nonlinear robust controller is complex and challenged for vehicle trajectory-following system, the main advantage of the proposed controller is in allowing application of powerful linear synthesis concepts to nonlinear vehicle dynamics system. In addition, compared with optimal control such as model predictive control, perhaps the proposed nonlinear robust control method is not better than optimal control in terms of optimal performance and implementation [11,12,23–26], whereas the proposed controller excels at robust stability and robust performance for the trajectory following system of AGEV where parameter uncertainties, system nonlinearity and external disturbances can be effectively dealt with using the proposed technique.

4. Simulation and analysis

In this section, the nonlinear robust H-infinity state-feedback controller (NRC) law designed above is simulated and verified on the Simulink-Carsim[®] joint platform. Simulink provides the overall framework environment, and the high-fidelity CarSim[®] software provides the dynamics model of AGEV trajectory following. The specific flowchart of system simulation framework is shown in Figure 2, the main parameters of AGEV are defined in Table 1. During the movement of the vehicle, system parameter uncertainty is mainly considered that the tire cornering stiffness of the N_{af} , N_{ar} will changes due to factors such as vehicle steering, ground roughness and environmental disturbances. Note that the tire cornering stiffness is variable and bounded, and the range of tire cornering stiffness in the system simulation: $N_{af} \in [79351, 96985] \text{ N/rad}$, $N_{ar} \in [97996, 119772] \text{ N/rad}$. Since the vehicle will inevitably encounter interference from the external environment during the movement, the interference often affects the change of vehicle state parameters, so that there is a certain difference between the measured signal and the actual signal. This paper assumes that the system interference $w = [0.01\sin(t), 0, 0.01\sin(t), 0]$. The simulated road conditions are set to double lane change (DLC) and serpentine curve scenes, and the forward speed is 72 km/h. The reference trajectory of DLC and serpentine curve are generated by a quintic polynomial technique, it is composed of a series of time-related vehicle state trajectory points, which contains information on vehicle position, heading angle, velocity, acceleration and radius of curvature. The reference trajectory we planned has already satisfied the obstacle avoidance function, because our work is not focused on trajectory planning of AGEV, here we assume that the reference trajectory we planned has already satisfied the obstacle avoidance function. For the dynamic trajectory planning, interested readers can refer to our previous and other related works [42,43]. In order to reflect the strong tracking ability and good steady-state response performance of the proposed controller, the linear quadratic regulator (LQR) controller and robust H-

infinity state-feedback controller (RHC) are also researched and compared.

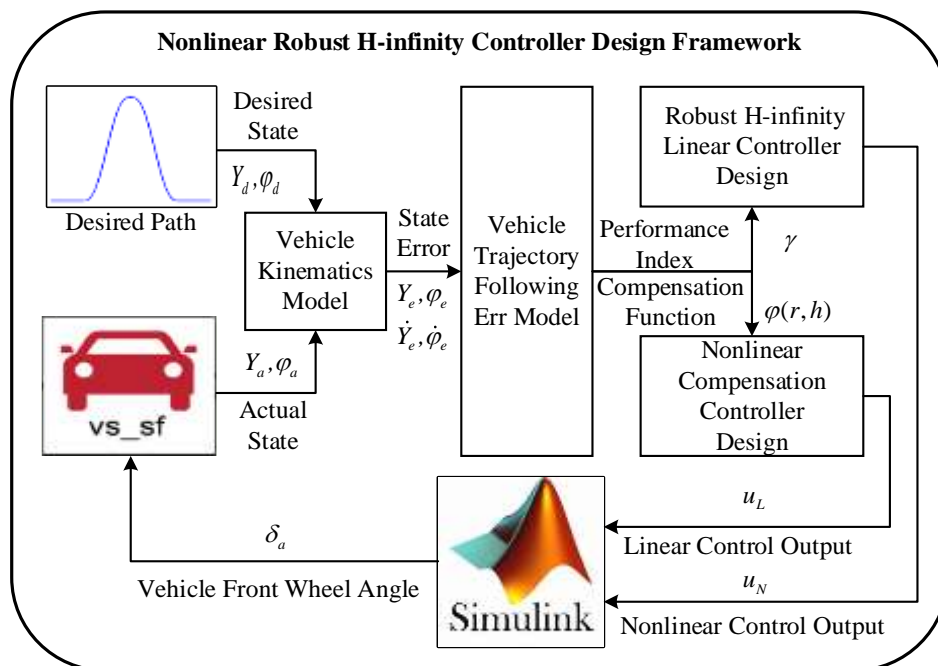


Figure 2. Specific flowchart of system simulation framework.

Table 1. Definition of key parameters of the vehicle.

Parameters	Explanation	Value(unit)
m	Vehicle mass	1413(kg)
I_z	The moment of inertia around the z -axis	1536.7(kg·m ²)
l_f	Distance from center of mass to front axle	1.015(m)
l_r	Distance from center of mass to rear axle	1.895(m)
N_{af}	Cornering stiffness of front wheels	88168(N/rad)
N_{ar}	Cornering stiffness of vehicle rear wheels	108884(N/rad)
l_z	Height of centroid to ground	0.54(m)
r	Wheel radius	0.325(m)

4.1. Double lane change condition

The simulation results of DLC including the global trajectories and lateral errors, curvature of road and the front wheel angle, yaw error, the lateral error and nonlinear compensation are shown in Figures 3–6. Figure 3 shows the global trajectories and lateral errors of different controllers under double lane change working conditions, among which the three controllers all have good tracking effects. The maximum lateral error of the LQR and RHC controller are about 0.4 m and 0.2 m respectively. The maximum lateral error of NRC is less than the other two controls, it indicates that the tracking performance of NRC is better than LQR and NRC. In addition, it can be seen from Figure 3, global X is between 56 and 65 meters, NRC also possesses outstanding system response when tracking DLC, which shows that proposed controller can improve the transient performance of

the closed-loop system.

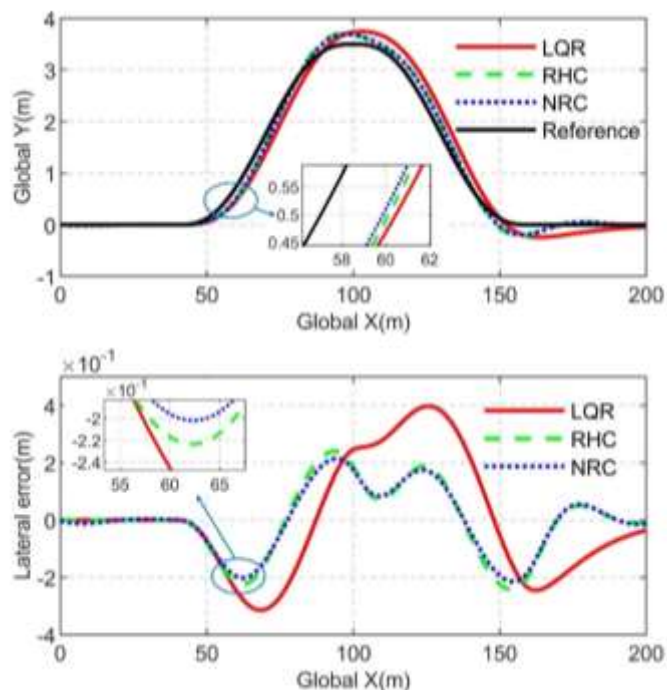


Figure 3. The global trajectories and lateral errors under DLC condition.

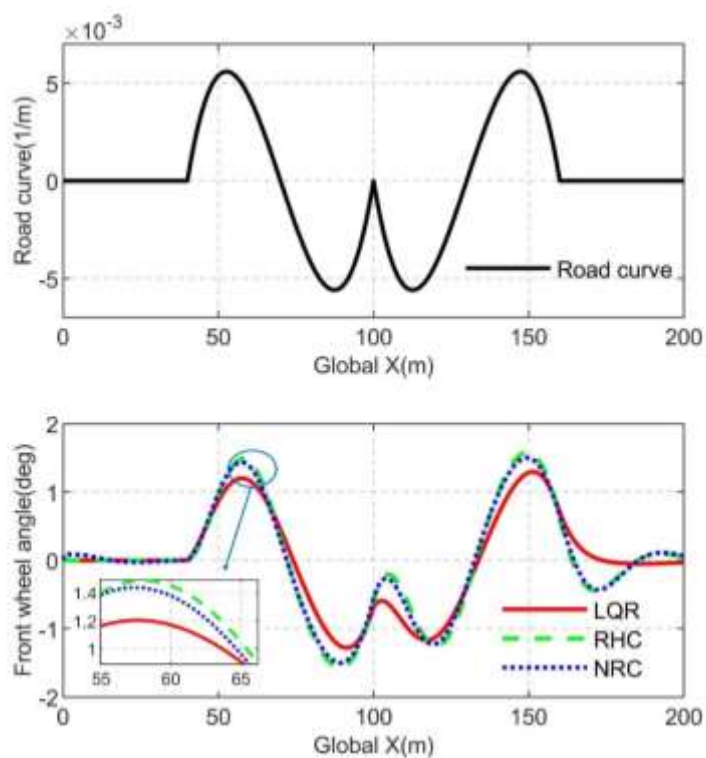


Figure 4. curvature of road and the front wheel angle under DLC condition.

Figure 4 shows the curvature of road and the front wheel angle of AGEV during tracking DLC. It can be seen from Figure 4 that the value of front wheel angle of NRC is always between LQR and RHC, which is because too small front wheel angle will lead to a small response speed of the system, while too large front wheel angle will lead to a large overshoot. However, the NRC has linear gain part to accelerate the system response, while the nonlinear compensation part can reduce large overshoot that results in excellent trajectory-following capability.

In Figure 5, it can be seen that NRC has smaller yaw error and stronger following ability compared with LQR and RHC. Figure 6 shows the magnitude of the lateral error and nonlinear compensation of NRC. It can see that when the lateral error is small, the nonlinear compensation part of NRC is large, and when the lateral error is increased, the nonlinear compensation part of NRC will become smaller, which indicates that the nonlinear compensation part meets the original intention of our NRC design, specifically the system has a fast response speed in the case of increasing error, and the system has a small overshoot under the condition of small error.

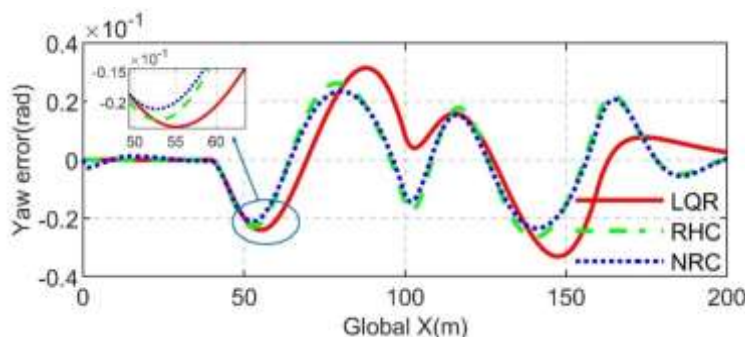


Figure 5. Yaw error under DLC condition.

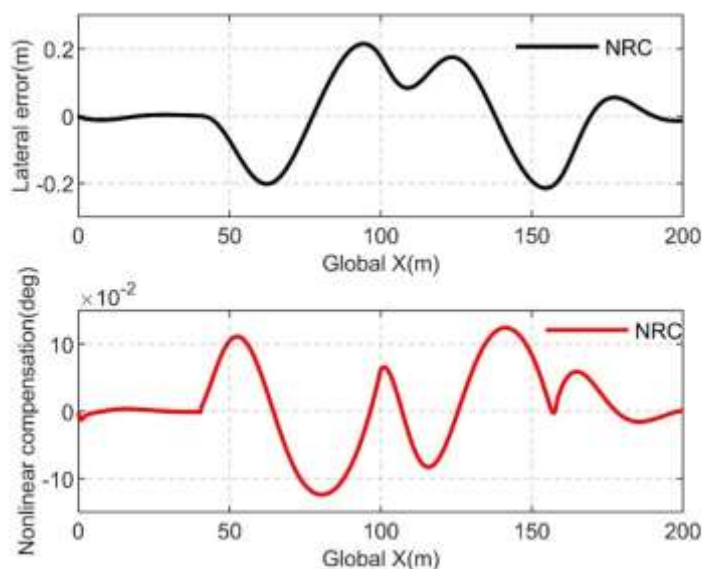


Figure 6. The lateral error and nonlinear compensation under DLC condition.

4.2. Serpentine condition

The simulation results of serpentine condition including the global trajectories and lateral errors, curvature of road and the front wheel angle, yaw error, the lateral error and nonlinear compensation are shown in Figures 7–10. Figure 7 shows the global trajectories and lateral errors of following in serpentine condition. The maximum lateral errors of NRC are smaller than those of LQR and RHC, and its response speed and transient performance are also higher than those of the other two controllers, it indicating that NRC has better tracking performance on serpentine roads than LQR and NRC.

Figure 8 shows the curvature and the front wheel angle of AGEV during tracking serpentine road. The maximum curvature of the serpentine road is 0.004. Similar to DLC condition, the value of front wheel angle of NRC is larger than LQR and smaller than RHC in serpentine working condition. Under the compensation of nonlinear part, NRC also has stable trajectory following ability.

In Figure 9, it can be seen that when AGEV follows a trajectory with large curvature, NRC can respond quickly with small yaw error, which brings low yaw error to the system and has stable tracking ability. Figure 10 displays the lateral error and nonlinear compensation under the serpentine condition, the maximum lateral error is 0.1 m. When the error changes, the nonlinear compensation part changes correspondingly, and it has the characteristics of fast response and small overshoot, and meets the requirements of trajectory following.

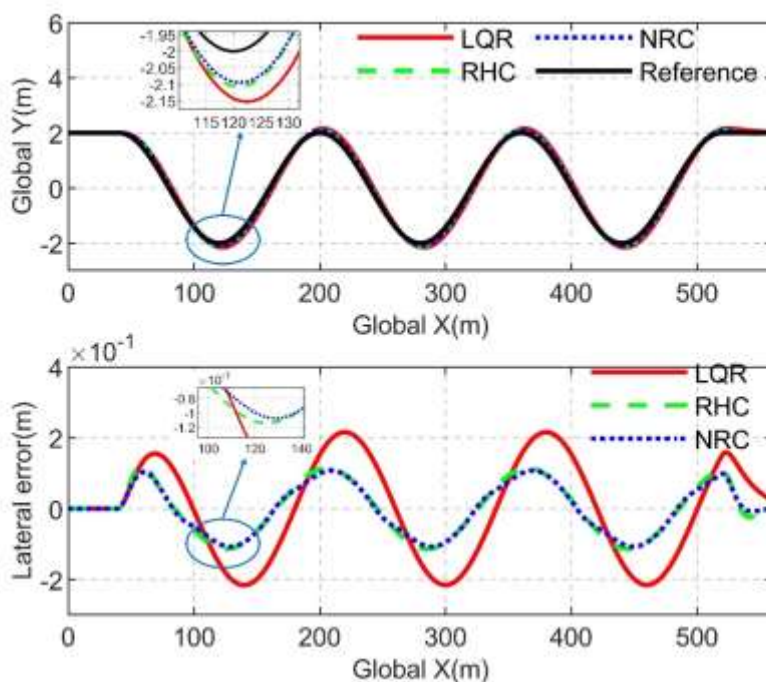


Figure 7. The global trajectories and lateral errors under serpentine condition.

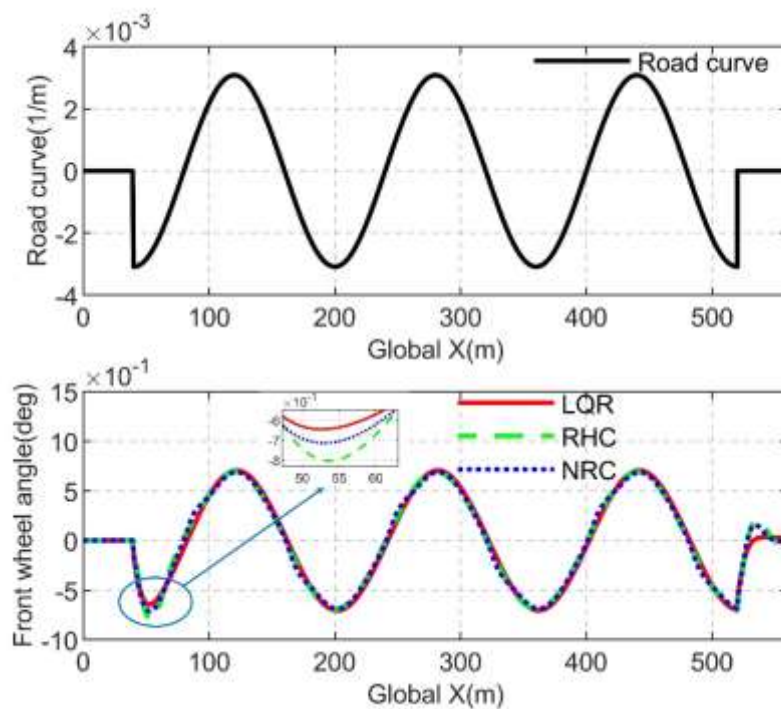


Figure 8. curvature of road and the front wheel angle under serpentine condition.

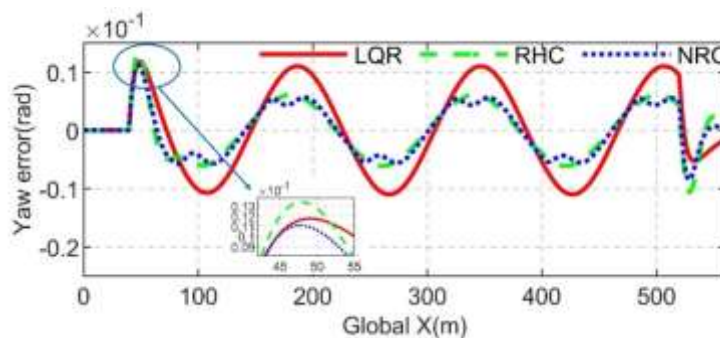


Figure 9. Yaw error under serpentine condition.

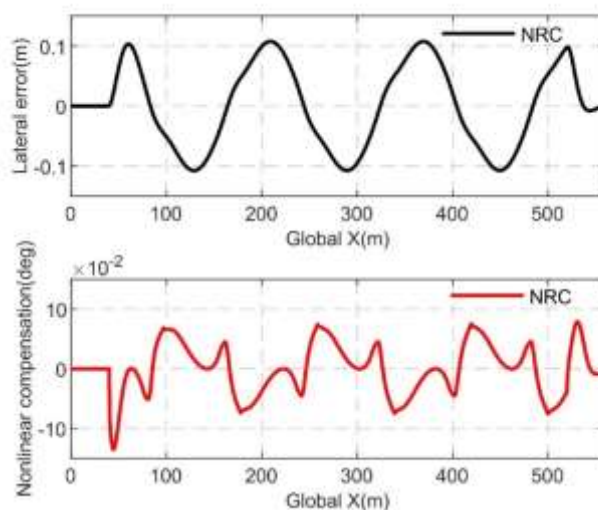


Figure 10. Lateral error and nonlinear compensation under serpentine condition.

For quantitative analysis of the tracking performance of the proposed controller, the maximum error (ME), the mean absolute error (MAE) and the root mean square error (RMSE) of the vehicle lateral displacement are used as the metrics for analysis, and the LQR and RHC are used as comparative experiments. The ME, MAE and RMSE are calculated as follows:

$$ME = \text{Max}\{|Y_{ai} - Y_{di}|, i = 1, 2, \dots, N\}, \quad (4.1)$$

$$MAE = \frac{\sum_{i=1}^N |Y_{ai} - Y_{di}|}{N}, \quad (4.2)$$

$$RMSE = \frac{\sqrt{\sum_{i=1}^N (Y_{ai} - Y_{di})^2}}{N} \quad (4.3)$$

where Y_{ai} and Y_{di} represent actual and ideal values, N is the number of samples.

Table 2 shows ME, MAE, RMSE, Reduction I and Reduction II under the lateral displacement of DLC and serpentine conditions. It can be seen from the data in Table 2 that NRC have smaller the ME, MAE and RMS than LQR and RHC in the two scenarios. From the point of view of working conditions, the error of DLC is greater than that of serpentine condition, which is due to the large lateral displacement of DLC, which leads to a larger error in trajectory following. In DLC condition, the overall improvement degree of NRC compared with LQR is more than 40%, and in serpentine condition, the degree of improvement of NRC compared with LQR is more than 50%. The ME of NRC is 10% higher than RHC under DLC condition, it indicating that NRC has a faster response speed than RHC in trajectory following with large errors. At the same time, the MAE of NRC is about 7% higher than that of RHC under DLC scenario, which shows that NRC has a smaller overshoot in trajectory following with small errors than RHC. On the whole, compared with LQR and RHC, the proposed controller has the advantages of fast response speed and small overshoot.

Table 2. Index analysis of lateral error under DLC and serpentine conditions.

Condition	Index(m)	LQR	RHC	NRC	Reduction I	Reduction II
DLC	ME	0.3977	0.2414	0.2146	46.04%	11.10%
	MAE	0.1538	0.0921	0.0859	44.15%	6.73%
	RMSE	0.1987	0.1248	0.1136	42.83%	8.97%
Serpentine	ME	0.2160	0.1211	0.1077	50.14%	11.07%
	MAE	0.1173	0.0629	0.0580	50.55%	7.79%
	RMSE	0.1372	0.0744	0.0684	50.15%	8.06%

Notes: Reduction I = $(LQR - NRC) / LQR$, Reduction II = $(RHC - NRC) / RHC$.

5. Conclusions

To improve the tracking accuracy, response speed and the overshoot suppression of the trajectory following control system, the nonlinear robust H_∞ state-feedback controller strategy of AGEV with AFS system is proposed. Firstly, the AGEV system dynamics and its control-oriented vehicle trajectory following system with dynamic error is developed. According to Lyapunov stability theory, the nonlinear robust H_∞ state-feedback controller of AGEV trajectory-following system is finally designed, solved by applying a set of linear matrix inequalities. Simulation including double lane change and serpentine conditions is implemented to verify the effectiveness of the proposed controller by using Matlab/Simulink-Carsim[®]. The simulation shows that the designed controller possesses effectual trajectory following performance of AGEV with AFS system. In addition, because we have compared the proposed method with LQR control and robust control in simulations, also the main purpose of this paper is to solve the engineering application problem of AGEV trajectory following control rather than pure control theory development. Therefore, we did not provide more robust optimal control and adaptive control as comparison. In the future, we will investigate and compare advanced robust optimal control and adaptive optimal control techniques such as SMC for trajectory following control for AGEV.

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Conflict of interest

The authors declare that there are no conflict of interests.

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